



## ⑥ origin point

$O_0 O_1 = v_{01}$  in tabella prendi i valori e moltiplichi  
( $i_{n-1} e k_n$ )

⑦ velocity of the origin  $O_i$ 

$$V_n = V_{n-1} + \omega_{n-1} \times O_{n-1} O_n + \dot{q}_n \delta_n k_n.$$

$$q_E = 0$$

$$\delta_E = a$$

## ⑧ Angular acceleration:

$$\dot{\omega}_i = \dot{\omega}_{i-1} + \ddot{q}_i (1 - \delta_i) \bar{K}_i + \dot{q}_i (1 - \delta_i) \bar{\omega}_{i-1} \times \bar{K}_i$$

## ⑨ linear acceleration

$$\begin{aligned} \dot{V}_i &= \dot{V}_{i-1} + \dot{\omega}_{i-1} \times (O_{i-1} O_i) + \bar{\omega}_{i-1} \times [\bar{\omega}_{i-1} \times (O_{i-1} O_i)] \\ &+ 2 \dot{q}_i \delta_i \bar{\omega}_{i-1} \times \bar{K}_i \end{aligned}$$

Position and rotation:

$${}^2\hat{A}_0 = ({}^0\hat{A}_1)^{-1} = \begin{bmatrix} {}^0A_1^T & -{}^0A_1^T \cdot {}^0p_1 \\ [0] & 1 \end{bmatrix}$$

translation:

$$\text{Tras}(\underline{s}) = \begin{bmatrix} 1 & 0 & 0 & s_x \\ 0 & 1 & 0 & s_y \\ 0 & 0 & 1 & s_z \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

In the global system:

$${}^0\hat{B}_1 = \text{Tras}(s) \cdot {}^0\hat{A}_1 = \begin{bmatrix} {}^0A_1 & \bar{p} + \bar{s} \\ [0] & 1 \end{bmatrix}$$

In the local system:

$${}^0\hat{B}_1 = {}^0\hat{A}_1 \cdot \text{Tras}(s) = \begin{bmatrix} {}^0A_1 & {}^0A_1 \bar{s} + {}^0\bar{p} \\ [0] & 1 \end{bmatrix}$$

# ROTATION MATRIX

about [x]

$$\text{Rot}(x, \theta) = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & c\theta & -s\theta & 0 \\ 0 & s\theta & c\theta & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

about [y]

$$\text{Rot}(y, \theta) = \begin{bmatrix} c\theta & 0 & s\theta & 0 \\ 0 & 1 & 0 & 0 \\ -s\theta & 0 & c\theta & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

about [z]

$$\text{Rot}(z, \theta) = \begin{bmatrix} c\theta & -s\theta & 0 & 0 \\ s\theta & c\theta & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

About the global:

$${}^0\hat{B}_2 = \text{Rot}(x, \theta) {}^0A_2 = \begin{bmatrix} \text{Rot}(x, \theta) \cdot {}^0A_2 & \text{Rot}(x, \theta) \cdot {}^0\bar{p} \\ [0] & 1 \end{bmatrix}$$

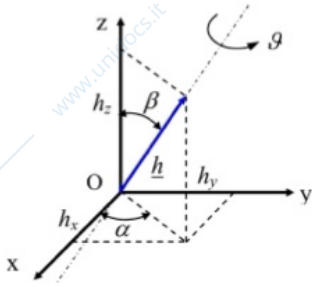
N.B.  
body = local

Fixed = global

About the local:

$${}^0\hat{B}_2 = {}^0A_2 \cdot \text{Rot}(x, \theta) = \begin{bmatrix} {}^0A_2 \cdot \text{Rot}(x, \theta) & {}^0\bar{p} \\ [0] & 1 \end{bmatrix}$$

# About an arbitrary Axis



$$R_{\hat{h}}(h, \vartheta) = \begin{bmatrix} h_x^2(1 - \cos \vartheta) + \cos \vartheta & h_x h_y(1 - \cos \vartheta) - h_z \sin \vartheta & h_x h_z(1 - \cos \vartheta) + h_y \sin \vartheta & 0 \\ h_x h_y(1 - \cos \vartheta) + h_z \sin \vartheta & h_y^2(1 - \cos \vartheta) + \cos \vartheta & h_y h_z(1 - \cos \vartheta) - h_x \sin \vartheta & 0 \\ h_x h_z(1 - \cos \vartheta) - h_y \sin \vartheta & h_y h_z(1 - \cos \vartheta) + h_x \sin \vartheta & h_z^2(1 - \cos \vartheta) + \cos \vartheta & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

I've nse problem

$$\theta = \cos^{-1} \left( \frac{R_{11} + R_{22} + R_{33} - 1}{2} \right)$$

if  $\sin \theta \neq 0$

$$\vec{h} = \frac{1}{2 \sin \theta} \begin{bmatrix} R_{32} - R_{23} \\ R_{13} - R_{31} \\ R_{21} - R_{12} \end{bmatrix}$$

$R_{32}$   $\rightarrow$  colonne  
 $\downarrow$   
 Righe =

if  $\sin \theta = 0$

$$h_x = \pm \sqrt{\frac{R_{11} + 1}{2}}$$

$$\text{sgn}(h_x h_y) = \text{sgn}(R_{21})$$

$$h_y = \pm \sqrt{\frac{R_{22} + 1}{2}}$$

$$\text{sgn}(h_x h_z) = \text{sgn}(R_{31})$$

$$h_z = \pm \sqrt{\frac{R_{33} + 1}{2}}$$

$$\text{sgn}(h_y h_z) = \text{sgn}(R_{32})$$

# Euler Angles

body - Axes  $z \times z - 313$

$$C = \mathcal{V} \text{Rot}(z, \vartheta) \text{Rot}(x, \vartheta) \text{Rot}(z; \psi)$$

$$= \begin{bmatrix} c\varphi \cdot c\psi - s\varphi \cdot c\vartheta \cdot s\psi & -c\varphi \cdot s\psi - s\varphi \cdot c\vartheta \cdot c\psi & s\varphi \cdot s\vartheta \\ s\varphi \cdot c\psi + c\varphi \cdot c\vartheta \cdot s\psi & -s\varphi \cdot s\psi + c\varphi \cdot c\vartheta \cdot c\psi & -c\varphi \cdot s\vartheta \\ s\vartheta \cdot s\psi & s\vartheta \cdot c\psi & c\vartheta \end{bmatrix}$$

if  $\sin \vartheta \neq 0$

$$\vartheta = \cos^{-1}(R_{33})$$

$$\varphi = \left( \frac{R_{13}}{s\vartheta} ; \frac{R_{23}}{s\vartheta} \right)$$

$$\psi = \text{Atan2} \left( \frac{R_{31}}{s\vartheta} ; \frac{R_{32}}{s\vartheta} \right)$$

if  $\sin \vartheta = 0$

$$\vartheta = 0^\circ (R_{33} = +1) \Rightarrow \varphi + \psi = \text{Atan2}(R_{21}, R_{11})$$

$$\vartheta = 180^\circ (R_{33} = -1) \Rightarrow \varphi - \psi = \text{Atan2}(R_{21}, R_{11})$$

## Angular velocity

$$\begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} = \begin{bmatrix} 0 & c\varphi & s\varphi \cdot c\vartheta \\ 0 & s\varphi & -c\varphi \cdot c\vartheta \\ 1 & 0 & c\vartheta \end{bmatrix} \begin{bmatrix} \dot{\varphi} \\ \dot{\vartheta} \\ \dot{\psi} \end{bmatrix}$$

FIXED - AXES XYZ

$$C = \text{Rot}(x, \varrho) \text{Rot}(y, \alpha) \text{Rot}(x, \psi) \cup$$

$$= \begin{bmatrix} c\varphi \cdot c\vartheta & c\varphi \cdot s\vartheta \cdot s\psi - s\varphi \cdot c\psi & c\varphi \cdot s\vartheta \cdot c\psi + s\varphi \cdot s\psi \\ s\varphi \cdot c\vartheta & s\varphi \cdot s\vartheta \cdot s\psi + c\varphi \cdot c\psi & s\varphi \cdot s\vartheta \cdot c\psi - c\varphi \cdot s\psi \\ -s\vartheta & c\vartheta \cdot s\psi & c\vartheta \cdot c\psi \end{bmatrix}$$

if  $\cos \vartheta \neq 0$

$$\vartheta = \sin^{-1}(-R_{31})$$

$$\psi = \text{Atan2}\left(\frac{R_{32}}{c\vartheta}, \frac{R_{33}}{c\vartheta}\right)$$

$$\varphi = \text{Atan2}\left(\frac{R_{21}}{c\vartheta}, \frac{R_{11}}{c\vartheta}\right)$$

if  $\cos \vartheta = 0$

$$\vartheta = 90^\circ (R_{31} = -1) \Rightarrow \varphi - \psi = \text{Atan2}(R_{12}, R_{22})$$

$$\vartheta = 270^\circ (R_{31} = 1) \Rightarrow \varphi + \psi = \text{Atan2}(-R_{12}, R_{22})$$

# FORMULARY

## NMS PART A

### ROD ELEMENT

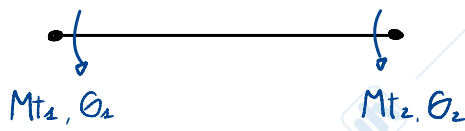


$$K_{rod} = \frac{EA}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

### THERMAL EXPANSION

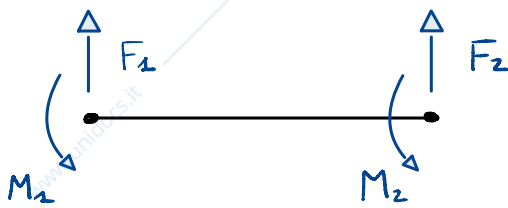
$$\Delta l = l \cdot \alpha \cdot \Delta T_m$$

### TORSION ELEMENT



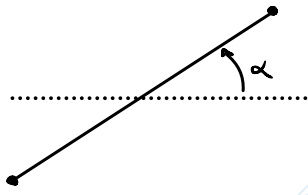
$$K_t = \frac{4J_t}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

### BEAM ELEMENT



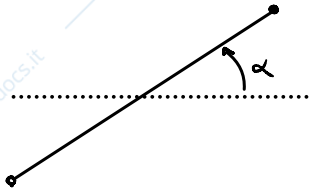
$$K = \begin{bmatrix} \frac{12EJ}{l^3} & \frac{6EJ}{l^2} & \frac{-12EJ}{l^3} & \frac{6EJ}{l^2} \\ \frac{6EJ}{l^2} & \frac{4EJ}{l} & \frac{-6EJ}{l^2} & \frac{2EJ}{l} \\ \frac{-12EJ}{l^3} & \frac{-6EJ}{l^2} & \frac{12EJ}{l^3} & \frac{-6EJ}{l^2} \\ \frac{6EJ}{l^2} & \frac{2EJ}{l} & \frac{-6EJ}{l^2} & \frac{4EJ}{l} \end{bmatrix}$$

## ROD ELEMENT ROTATION



$$[T] = \begin{bmatrix} c\alpha & s\alpha & 0 & 0 \\ 0 & 0 & c\alpha & s\alpha \end{bmatrix}$$

## BEAM ELEMENT ROTATION



$$[T] = \begin{bmatrix} c\alpha & -s\alpha & 0 & 0 & 0 & 0 \\ s\alpha & c\alpha & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & c\alpha & -s\alpha & 0 \\ 0 & 0 & 0 & s\alpha & c\alpha & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$